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A Comparison of Rivet Shear Strengths Obtained from Two-Rivet Specimens and from Multi-Rivet Specimens at Room Temperature and 150°C

by

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A COMPARISON OF RIVET SHEAR STRUNGTHS OBTAINED FROM TWO-RIVET SPECIMENS AND FROM MULTI-RIVET SPECIMENS AT ROOM TEMPERATURE AND 150°C

by

D. F. Wright P. Judson, B.Sc.(Eng.). and P. W. Horrocks

SUMMARY

Comparative tests were made to investigate the strength per rivet obtained in a two-rivet tandem lap joint specimen and a multi-rivet lap joint specimen approximating to a realistic joint as used in aircraft. Tests were made on snaphead-riveted aluminium alloy specimens at room temperature, with and without prior soaking at temperature, and also at an elevated temperature of 150° C with prior soaking. The soaking condition used was 150° C for 1000 hours.

The tests showed that there is little difference in strength per rivet between the two types of specimen, but heat factors obtained for the proof conditions on the two-rivet specimen might be optimistic. No corrosion protection was applied and the results of these tests might be modified by the addition of an interlayer compound.

Replaces R.A.E. Tech Note No. Structures 360-A.R.C. 26393.

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1 INTRODUCTION

This paper gives the results of tests made to compare the static shear strength of snaphead rivets in a simple riveted lap joint specimen having two rivets in tandem, with that of rivets in a multi-rivet lap joint more representative of actual aircraft construction. The purpose of the tests was to check that design strength values obtained from the simpler two-rivet-in-tandem type of lap joint in aluminium alloy were acceptable.

The tests were made:-

(a) at room temperature, both with and without prior soaking of the specimens at 150° C for 1000 hours

(b) at 150°C, with prior soaking of the specimens at 150°C for 1000 hours.

The tests showed that rivet strengths in static shear obtained from the two-rivet lap-joint specimens did not differ significantly from those obtained from the multi-rivet lap joints, at both room temperature and 150°C, but that heat factors* obtained for the proof conditions on the two-rivet specimen might be optimistic.

2 TEST SPECIMENS AND TEST PROGRAMME

The two-rivet specimen is shown in Fig. 1, and the multi-rivet specimen in Fig. 2. Both types of specimen were constructed from 16 S.W.G. aluminium alloy sheet material to B.S. L.73, and 5/32" diameter aluminium alloy snaphead rivets to B.S. S.P.80 (B.S. I.86 rivet wire). Extracts from these specifications are given in Appendix 1.

Thirty six specimens of each type were prepared in such a way that the direction of test loading was transverse to the final rolling direction of the sheet material. No corrosion protection was applied to the joints before assembly.

The test programme is given in Table 1.

3 TEST LQUIPMENT

3.1 Test machines

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The two-rivet specimens were tested at room temperature on the 0-3000 lb range of a 15,000 lb Denison test machine and at elevated temperature on the 0-2000 lb range of a 10,000 lb Mohr and Federhaff test machine.

The multi-rivet specimen tests at room and elevated temperatures were made on the 0-10 ton range of a 50 ton Avery test machine. All test machines were to Grade A of B.S.1610.

\$ Heat Paston		Strength af	ter expo	<u>sure t</u>	o elevated	temperature
neat lactor	=	Str	ength at	room	temperature	; · · · · · · · · · · · · · · · · · · ·

3.2 Furnace equipment for hot tests

The two-rivet joint specimens were tested in a Davall furnace, approximately $1\frac{1}{2}$ " x 3" working cross-section and 8.5" deep. This furnace had separate heaters for the top and bottom halves, each separately controlled by a variable trans-former. The gaps between the furnace and the specimen were plugged with asbestos in such a way that convection currents were reduced without introducing constraint on the specimen.

The multi-rivet specimens were tested in a "book" type furnace, fitted with two separate pairs of heating elements. The current to each pair could be independently adjusted. The furnace had slots on each side to accommodate the extensometer clamps. These slots were not sealed, but the gaps between the furnace and the specimen were plugged in the same way as for the two-rivet specimens.

3.3 Extensometers

The extension of the two-rivet specimens at room temperature was measured by an R.A.E. dial extensometer, with the dial gauge reading to 0.0001"; and at elevated temperature, a modified Lamb-type optical roller extensometer was used.

The extension of the multi-rivet specimen was measured by a special extensometer consisting of four clamps, two on each side of the specimen, spanning the joint at the edge of the sheet (Fig. 3). Dial gauges (reading to 0.0001") between each pair of clamps registered the extensions on each edge of the specimen. The mean of the readings from each side was taken to be the extension of the specimen. The validity of this assumption was investigated on a few specimens by making additional comparative measurements (with a Southern Instruments Limited "Magna Gage" inductance transducer) of the relative movement of two small brackets attached at the centre line, on one side of the specimen, one on each side of the lap.

4 METHOD OF TEST

4.1 Loading and extension

For both room and elevated temperature tests, the load was applied to each specimen in increments of 100 lb (reducing to 25 lb at high loads) for the tworivet specimens, and in increments of 0.1 ton for the multi-rivet specimens. Extension measurements were made at each increment of load until the load corresponding to a total permanent joint extension of 4% of the nominal rivet diameter (the r₂₀ proof value) had been exceeded. The load was then increased continuously until failure occurred, the failing load being recorded. The proof strengths were calculated on the assumption that the permanent set was equal to the offset of the load extension curve from the continuation of the linear part of the curve.

4.2 <u>Heating of the specimens</u>

Before any tests were made at elevated temperature, dummy specimens (not used in subsequent tests) were used to adjust the apparatus to the correct temperature distribution, as follows.

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4.2.1. Two-rivet specimens

Three thermocouples, one at each end of the lap joint and one in the centre of the lap, were attached to the specimen, each thermocouple wire being independently "welded" to the specimen on the centre line. (See Fig. 1.) The voltage outputs of the thermocouples were measured on a potentiometer and the readings converted to temperature. The current in the windings of the furnace was adjusted so that the temperature distribution as indicated by the above thermocouples, was within the limits $150^{\circ}C \pm 3^{\circ}C$. During the tests, the temperature of each specimen was checked by thermocouples attached at the same positions as were used on the dummy specimen. The time taken for each specimen to reach a stable temperature of $150^{\circ}C$ was about one hour.

4.2.2 Multi-rivet specimens

Five thermocouples, with each wire independently "welded" to the dummy specimen, were attached as follows. (See Fig. 2.)

(i) One thermocouple on the centre line of the specimen mid-way between the rivet rows.

(ii) Two thermocouples symmetrically placed about the centre line 4.5" apart, 2" above the rivets.

(iii) Two thermocouples symmetrically placed about the centre line 4.5" apart, 2" below the rivets.

As before, the voltage outputs of the thermocouples were measured on a potentiometer, the readings converted to temperature, and the currents in the furnace windings adjusted so that the temperature distribution, as indicated by the thermocouples, was within the limits of $150^{\circ}C \pm 3^{\circ}C$.

Each of the specimens used in the tests had three thermocouples attached, at the points shown in Fig. 2. These thermocouples were used to check the temperature of the specimens during test. The time for each specimen to reach a stable temperature of 150° C was about $1\frac{1}{4}$ hours.

5 DISCUSSION OF RESULTS

Table 2 gives the r_{10} and r_{20} proof loads and the ultimate load, R, for each specimen, and also the mean values and coefficients of variation for each group of results. Table 3 summarizes the results and gives the percentage difference in strength-per-rivet between the two-rivet and the multi-rivet joints.

Fig. 4 shows typical curves of load-per-rivet against extension for both types of specimen for each of the three test conditions. It may be noted that there is a marked difference in slope of these curves for the two types of specimen, due to the tension stress in the sheet portion of the specimen being higher in the multi-rivet specimens than in the two-rivet specimens. As, however, these tension stresses were below the elastic limit in all cases, and thus produced no permanent extension, this difference in slope does not affect the proof strength values obtained in the tests.

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Fig. 5 shows curves of load against extension at room temperature for the multi-rivet specimen, with extensions measured both by dial gauges indicating extension at each side of the specimen, and also by an inductance transducer measuring extension at the centre of the specimen. The mean extension obtained from the readings of the two dial gauges did not entirely agree with the extension shown by the inductance transducer, but the proof strengths derived from the two sets of measurements were in good agreement, so that it is considered that the readings of the dial gauges are valid for the purposes of this investigation.

Fig. 6 is a pictorial summary of the results and shows that in general for each test condition the difference between the strengths-per-rivet obtained from the two types of specimen is small.

At room temperature, with no prior heat soaking, the multi-rivet specimen was the stronger, particularly for the proof conditions. After heating for 1000 hours at 150°C, the tests at room temperature showed that there was little difference in strength between the two types of specimen but, for the proof conditions, the two-rivet specimen had higher heat factor values than the multirivet specimen. Testing at 150°C after a similar heating period gave a small strength advantage to the two-rivet specimen and, for the proof conditions, heat factors higher than those for the multi-rivet specimen. For both test conditions after prior heat soaking, the heat factors for the ultimate condition were similar for both types of joint. In all the tests the coefficients of variation were quite low for this type of test.

The strength differences between the two types of joint were small. However, the stronger proof values changing from one type of specimen to the other with exposure to elevated temperature had a more marked effect on the heat factors. A possible explanation is that at room temperature without prior heat soaking, there is more clamping between the two sheets of the multi-rivet specimen giving enhanced proof values but not affecting the ultimate strength. There is a reduction in this clamping pressure after exposure to elevated temperature, possibly due to creep in tension of the rivets. This is more marked in the case of the multi-rivet specimens thereby giving lower heat factors than on the two-rivet specimen. The heat factors greater than unity for the proof conditions indicate that the sheet material as recieved was not fully aged. This is a normal condition of supply and does not affect the validity of the comparisons made between the two types of specimen.

6 CONCLUSIONS

The differences in strength-per-rivet between two-rivet and multi-rivet joint specimens for the test conditions investigated were small. For the tests involving prior heat soaking, the heat factors for the proof conditions were lower for the multi-rivet specimen than for the two-rivet specimen.

For the types of joint investigated the differences are sufficiently small to warrant the use of rivet shear strength data obtained from the tworivet type of joint in the design of multi-rivet joints within the range of temperature used in the tests. Presentation of these data in heat factor form could be optimistic. No corrosion protection was applied and the results of these tests might be modified by the addition of an interlayer compound. The strength values given in this Note should not be used as design strengths, as they were obtained from tests on material in an unknown agohardened condition, and may therefore be optimistic.

SYLIBOLS

R = Failing load per rivet* in 1b

r₁₀ = Proof load per rivet in 1b to give a permanent set of the whole joint equal to 2 per cent of the rivet diameter.

r₂₀ = Proof load per rivet in 1b to give a permanent set of the whole joint equal to 4 per cent of the rivet diameter.

 $\overline{\mathbf{x}}$ = Mean strength.

n = No. of results.

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v = Coefficient of variation = $\frac{1}{\overline{x}} \sqrt{\frac{\Sigma(x-\overline{x})^2}{(n-1)}}$

* = The load per rivet is taken as

 $= \frac{\text{Total load on the specimen}}{\text{Number of rivets in the joint}}$

APPENDIX 1

EXTRACTS FROM B.S. SPECIFICATIONS L.73 AND L.86

L.73

Aluminium coated aluminium - copper - magnesium - silicon - manganese alloy sheet and strips having the following chemical composition:-

Copper	not	less	than	3.8%	nor	more	than	4.8%
Magnesium	11	11	Ħ	0.55%	11	11	*1	0.85%
Silicon	11	11	н	0.6%	f t	11	n	0.9%
Iron	not	more	than	1.0%				
Manganese	not	less	than	0.4%	nor	more	than	1.2%
Nickel	not	more	than	0.2%				
Zinc	11	11	11	0.2%				
Lead	11	11	11	0.05%				
Tin	11	11	tt	0.05%				
Titanium and/or Chromium	н	13	11	0.3%				
Aluminium	the	remai	inder					

Solution treated at $505 \pm 5^{\circ}$ C and quenched in water not exceeding 40° C. Artificially aged at between 160° C and 190° C.

Mechanical properties:- 0.1% proof stress not less than 21 tons/sq in. (thicker than 25 S.W.G.) Ultimate stress not less than 27 tons/sq in. Elongation not less than 8%.

L.86

Aluminium - copper - magnesium alloy wires for solid, cold forged rivets having the following chemical composition:-

Copper	not	less	than	1.5%	nor	more	than	3.0%
Magnesium	11	11	11	0.2%	\$1	11	11	0.5%
Silicon	not	more	than	0.7%				
Iron	11	11	**	0.7%				
Manganese	11	Ħ	11	0.5%				
Nickel	11	n	11	0.2%				
Zinc	11	11	11	0.1%				
Lead	11	11	11	0.05%				
Tin ·	11	11	11	0.05%				
Titanium	15	н	11	0.3%				
Aluminium	the	remai	inder					

Supplied annealed and subsequently cold drawn to secure a reduction in crosssectional area of not less than 20% and not more than 40%.

Solution treated at $495 \pm 5^{\circ}$ C and quenched in water not exceeding 40° C. Naturally aged at room temperature for not less than 48 hours.

Mechanical properties: - Ultimate stress not less than 17 tons/sq in.

TABLE 1

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TEST PROGRAMME

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	TWO-RIVET SPECIMENS								
Specimen No.	Soaking	Soaking	Test						
	temperature	time	temperature						
1 - 12	150 ⁰ C	1000 hrs	150 ⁰ C						
13 - 24	150 ⁰ C	1000 hrs	Room temp						
25 - 36	No	ne	Room temp						
	MULTI-RIVET SPECIMENS								
1 - 12	150°C	1000 hrs	150 ⁰ C						
13 24	150 [°] C	1000 hrs	Room temp						
25 - 36	No	ne	Room temp						

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TABLE 2

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епр	Failure R	660 Ib 695 695 695 690 685 690 685 685 685 680 682 1•7	695 Ib 700 695 695 676 676 683 683 683 683 683 1•4 1•4
st at room t	Proof r ₂₀	582 Ib 620 582 610 575 585 585 588 588 588 588 588 588 588	605 Ib 635 620 624 624 624 600 625 602 602 602 608 617 Ib 2.3
soaking ter	Proof r ₁₀	525 Ib 550 550 550 535 535 535 535 535 535 535	569 1b 595 586 564 574 574 578 578 578 574 578 574 577 2.60
No heat	Specimen No.	びあびぬめみれだがれがあ	むあびゅめねれ び び 丼 ちお
	Failure R	670 Ib 650 Ib 655 655 655 655 655 655 655 655 655 1.3	660 Ib 660 660 671 660 640 638 676 676 676 644 644 2•3 Ib
r 1000 hrs temp	Proof r ₂₀	640 Ib 640 Ib 608 610 604 616 616 618 612 612 612 612 1.9 1.9	608 Ib 624 624 624 605 595 595 608 624 594 594 594 2•3 Ib
id 150 ⁰ C foi test room 1	Proof r ₁₀	586 Ib 568 Jb 560 5566 584 5566 578 Ib 578 Jb 578 Jb	585 1b 589 595 582 582 573 573 582 582 582 582 581 1b 1•9 1•9
Soake	Specimen No.	<u> </u>	545656 8 2 8 84
	Failure R	620 lb 540 537 557 550 550 555 555 5.7 lb 5.7 lb	520 520 520 525 525 525 543 543 525 525 525 517 523 517 523 517 523 517 523 517
at 150°C	Proof r ₂₀	p distbn readings 522 505 502 502 515 500 515 500 502 502 502 502 502 502 502 502 50	p distribution 476 lb 501 490 481 malfunction cold and at 481 481 481 487 lb 487 lb 1.6
1000 hrs test	Froof r ₁₀	Used for tem Insufficient 4.85 500 1nsufficient 500 4.90 4.90 4.87 4.87 4.86 4.92 4.92 4.87 1b 2.3 2.3	Used for tem 4448 lb 474 467 467 467 461 Extensometer Pre-stressed 150°C 459 Specimen ove Controller m 470 464 lb 464 lb
Soaked 150°C for	Specimen No.	1 22 44 66 10 11 12 Mean load per rivet W%	1 44 6 7 10 11 11 12 Mean load per rivet
		anemioeqa tevir-owT	anemiceqa tevir-itluM

V = Coefficient of variation

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TABLE 3

COMPARISON OF MEAN r₁₀ AND r₂₀ PROOF LOADS, AND FAILING LOADS R FOR THE TWO TYPES OF SPECIMEN

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	Proof load r10			Proof load r ₂₀			Failing load R			
Soaking temp	15	000	Not	150°C Not 1000 hours soaked		Not	150 ⁰ C 1000 hours		Not	
Soaking time	1000 1	hours	soaked			soaked			soaked	
Test temp	150°C	R.T.	R.T.	150 ⁰ C	R.T.	R.T.	150°C	R.T.	R.T.	
Two rivet specimen	487 (0.91)	578 (1.08)	537	508 (0.87)	615 (1.05)	587	539 (0.79)	659 (0.96)	684	
Multi-rivet specimen	464 (0.80)	581 (1.01)	577	487 (0 .79)	605 (0.98)	617	523 (0.76)	655 (0.95)	687	
Difference	+23	-3	-40	+21	+10	-30	+16	+4	-3	
Difference as a percentage of the mean strength per-rivet of the two-rivet specimens.	+4.7	-0 . 5	-7•4	+3.9	+1.6	 5 . 1	+3.0	+0.6	-0.4	

R.T. = Room temperature Figures in brackets are heat factors

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FIG. I. TWO-RIVET TEST SPECIMEN



FIG. 2. MULTI-RIVET TEST SPECIMEN



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FIG. 3. EXTENSOMETER ARRANGEMENT FOR MULTI-RIVET SPECIMENS

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FIG. 4. TYPICAL LOAD-EXTENSION CURVES FOR TWO-RIVET AND MULTI-RIVET SPECIMENS.

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FIG. 5 LOAD EXTENSION CURVES FOR MULTI-RIVET SPECIMEN Nº15 SHOWING MAGNA GAGE' CURVE.



FIG. 6. COMPARISON BETWEEN STRENGTHS FOR TWO-RIVET AND MULTI-RIVET SPECIMENS.

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A COMPARISON OF RIVET SHEAR STRENGTHS OBTAINED FROM TW RIVET SPECIMENS AND FROM MUTLI-RIVET SPECIMENS AT ROOM TEMPERATURE AND 150°C. Wright, D.F., Judson, P., and Horrocks, P.W. August 1964.	10-	A COMPARISON OF RIVET SHEAR STRENGTHS OBTAINED FROM TWO- RIVET SPECIMENS AND FROM MULTI-RIVET SPECIMENS AT ROCM TEMPERATURE AND 150°C. Wright, D.F., Judson, P., and Horrocks, P.W. August 1964.	· .
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